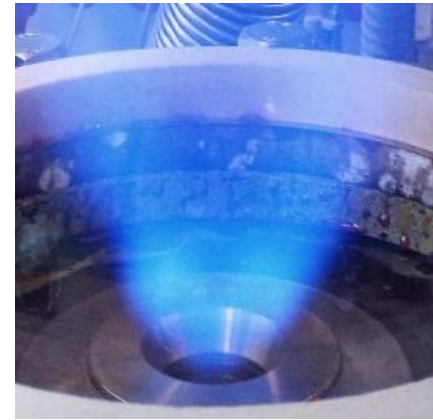
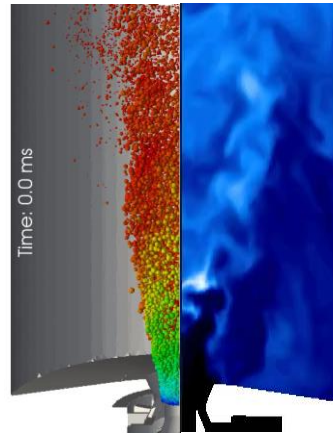
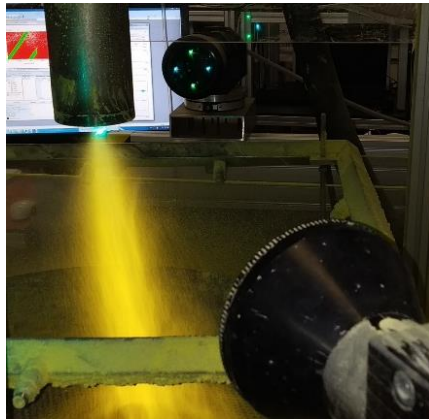


# Sulphur combustion at high power density

PEGASUS workshop 09/09/2021

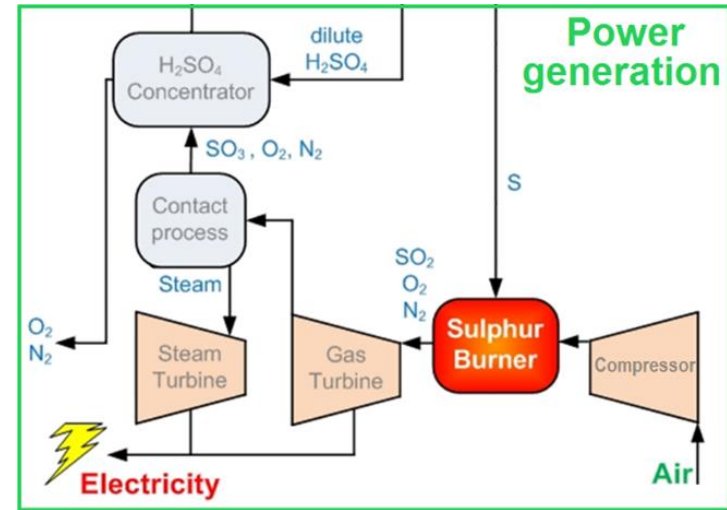
M. Fedoryk, N. Sebbar, S. Harth, D. Trimis





# Development of sulphur burner with high power density suitable for integration in gas turbine

- Definition of operating conditions for sulphur burner
  - Oxidator: air
  - Air compression ratio of gas turbine: 15 bar
  - Combustor air inlet temperature: 720 K
  - Thermal power:
    - 20 kW – laboratory scale (1 burner in tubular combustor, 1 bar)
    - 5 MW - full-scale (multiple burners in annular combustor, 15 bar)
    - Power density target  $>1.5 \text{ MW/m}^3$  (1 bar)
  - Turbine inlet temperature shall be in the range of 1530 K – 1700 K
    - Combustion in the burner to mainly  $\text{SO}_2$
    - Excess oxygen needed for the following contact process ( $\text{SO}_2 \rightarrow \text{SO}_3$ )
    - Preferred air/sulphur equivalence ratio  $\lambda \approx 2$  (min. 1.5)

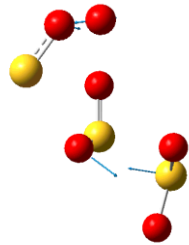


# Sulphur combustor development methodology

- Development of Sulphur combustor with **high power density**
- Development of **chemical kinetics mechanism** for Sulphur combustion
- **CFD simulations** to support burner development
- Design of laboratory test rigs for
  - **atomization** of Sulphur at isothermal condition
  - **combustion** of Sulphur
- Experimental investigation of elemental **Sulphur atomization**
- Experimental validation of lab-scale **Sulphur combustor**



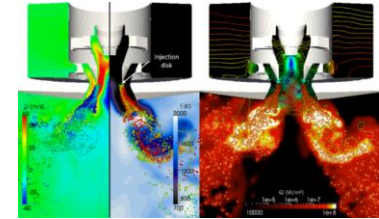
# Sulphur kinetics needed for further development steps



Combustion kinetics development



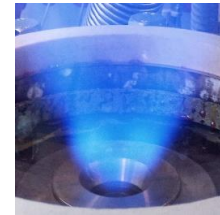
CFD Simulations



Sulphur Atomization (lab scale)



Sulphur Combustion (lab scale)



- **Kinetics – not available / validated for pure Sulphur**
- Flame speed for premixed condition (unknown)
- Auto ignition time (unknown)



# Sulphur combustion kinetics: Reactions from literature

$$k_r = AT^b \exp\left(-\frac{T_a}{T}\right)$$

REACTIONS	MOLES	KELVINS	A	b	T <sub>a</sub>
S+O2 = SO+O			5.200E+06	1.8100	-600.00
S2+M = 2S+M			4.800E+13	0.00	38800.00
S2+O = SO+S			1.000E+13	0.0000	0.00
SO3+O = SO2+O2			2.000E+12	0.0000	10000.00
SO3+SO = 2SO2			1.000E+12	0.0000	5000.00
SO+O (+M) = SO2 (+M)			3.200E+13	0.0000	0.00
N2/1.5/ SO2/10/ LOW /			1.200E+21	-1.54	0.00 /
TROE /			0.5500	1.0e-30	1e+30 /
SO2+O (+M) = SO3 (+M)			9.200E+10	0.0000	1200.00
LOW /			2.400E+28	-4.00	2640.00 /
SO+M = S+O+M			4.000E+14	0.0000	54000.00
N2/1.5/ SO2/10/					
SO+O2 = SO2+O			7.600E+03	2.3700	1500.00
2SO = SO2+S			2.000E+12	0.0000	2000.00
SO3+S = SO+SO2			5.120E+11	0.00	0.00

■ Numerical predictions with CHEMKIN

■ Target: Burning velocities

Extracted from a hydrocarbon combustion mechanism (ca. 430 react.) which includes treatment/oxidation of sulfur present in the system

[1] Hughes, K. J.; Blitz, M. A.; Pilling, M. J. Robertson, S. H.  
*Proc. Comb. Inst.* 29: 2431–2437 (2002).



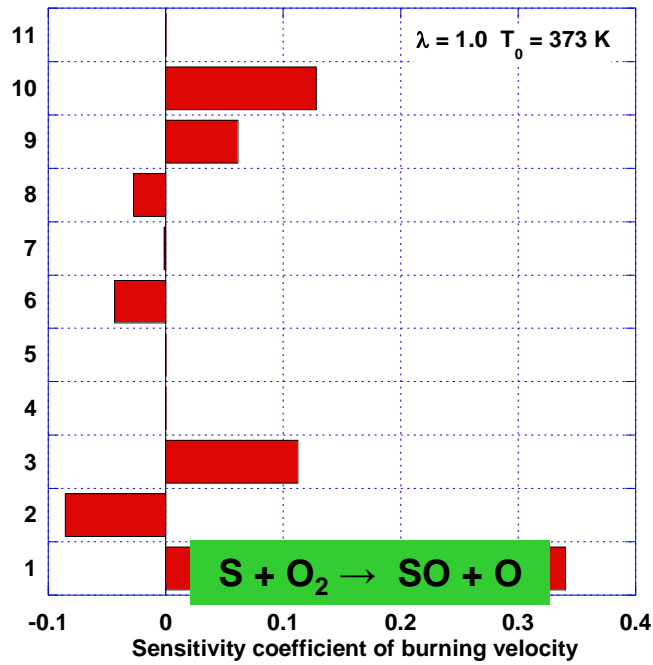
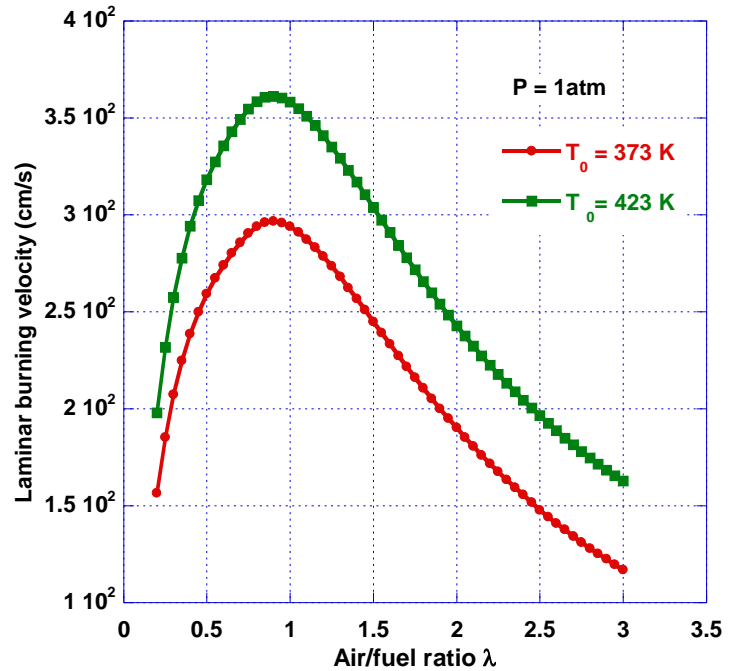
# Sulphur combustion kinetics: Laminar burning velocity

## Reactions from literature

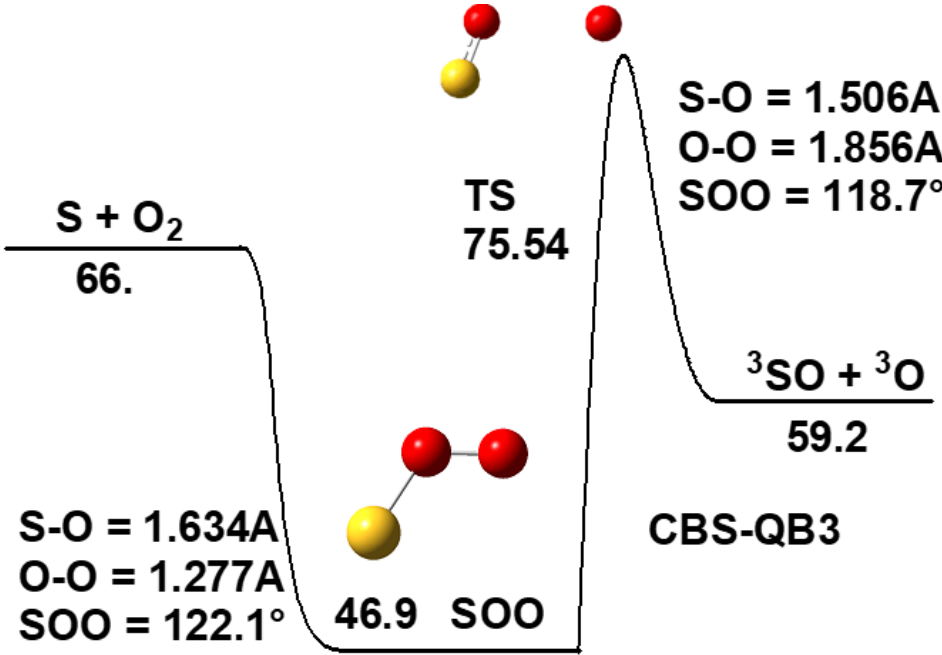
### Laminar burning velocity for $S_2 + O_2$

### Sensitivity analysis ( $\lambda=1$ )

With Literature mechanism: 11 Reactions



# Sulphur combustion kinetics: Own estimation of $S + O_2 \rightarrow SO + O$



Overall Reaction Rate from Master Equation Analysis	$k = A T^b \exp(-T_a/T)$		
	A	b	T <sub>a</sub>
$S + O_2 \rightarrow SO + O$ singlet	2.69E+10	0.15	5358.8
$k(1600\text{ K}) = 2.86E+09$			
$S + O_2 \rightarrow SO + O$ triplet	1.09E+11	0.15	2369.4
$k(1600\text{ K}) = 7.50E+10$			

N. Sebbar, T. Zirwes, P. Habisreuther, J.W. Bozzelli, H. Bockhorn, D. Trimis. *Energy & Fuels* **2018** 32 (10), 10184-10193



# Sulphur combustion kinetics: Modification of literature mechanism

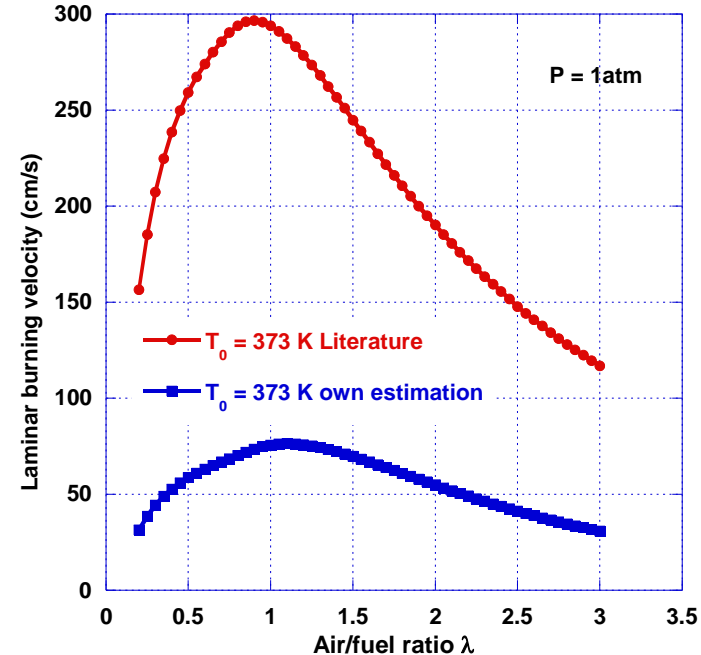
- lit. mechanism [1]  
„base mechanism“

REACTIONS	MOLES	KELVINS	A	b	T <sub>a</sub>
S+O2 = SO+O			5.200E+06	1.8100	-600.00
S2+M = 2S+M			4.800E+13	0.00	38800.00
S2+O = SO+S			1.000E+13	0.0000	0.00
SO3+O = SO2+O2			2.000E+12	0.0000	10000.00
SO3+SO = 2SO2			1.000E+12	0.0000	5000.00
SO+O(+M) = SO2(+M)			3.200E+13	0.0000	0.00
N2/1.5/ SO2/10/ LOW /			1.200E+21	-1.54	0.00 /
TROE /			0.5500	1.0e-30	1e+30 /
SO2+O(+M) = SO3(+M)			9.200E+10	0.0000	1200.00
LOW /			2.400E+28	-4.00	2640.00 /
SO+M = S+O+M			4.000E+14	0.0000	54000.00
N2/1.5/ SO2/10/ SO+O2 = SO2+O			7.600E+03	2.3700	1500.00
2SO = SO2+S			2.000E+12	0.0000	2000.00
SO3+S = SO+SO2			5.120E+11	0.00	0.00

- S + O<sub>2</sub> → SO + O is replaced  
by two reactions [2]  
“modified mechanism”

REACTIONS	MOLES	KELVINS	A	b	T <sub>a</sub>
S + O2 = SO + O			2.69E+10	0.15	5358.8
DUP					
S + O2 = SO + O			1.09E+11	0.15	2369.4
DUP					

## Significant impact of new kinetics on flame speed



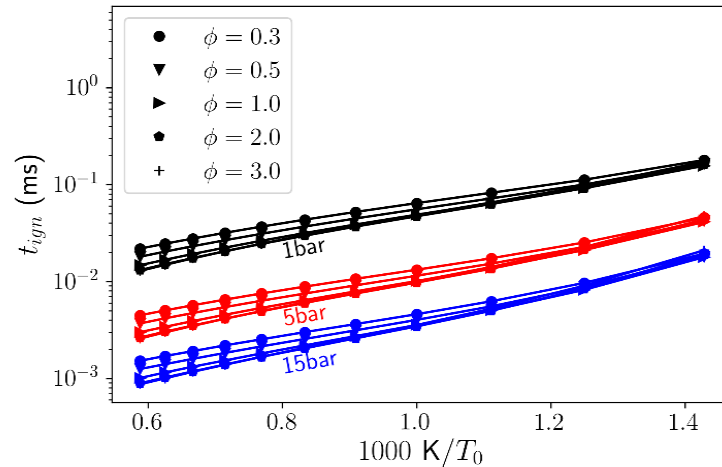
[1] Hughes, K. J.; Blitz, M. A.; Pilling, M. J. Robertson, S. H. *Proc. Comb. Inst.* 29: 2431–2437 (2002).

[2] Sebbar, N.; Zirwes, T.; Habisreuther, P. Bozzelli, J. W.; Bockhorn, H.; Trimis, D. *Energy Fuels* 2018, 32, 10184–10193

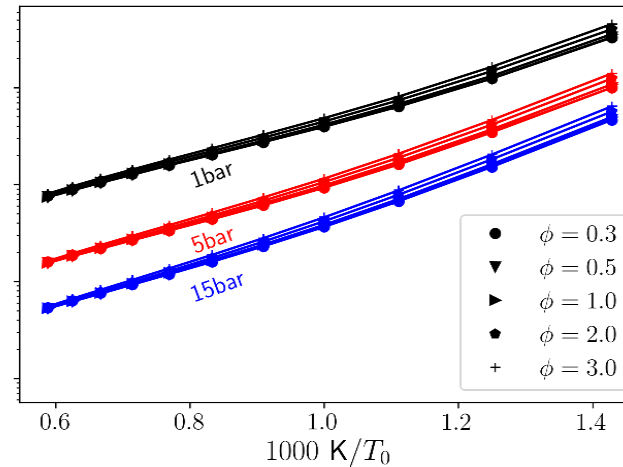
# Sulphur combustion kinetics: Ignition delay time

- In the modified mechanism, ignition delay times are about 5 times larger
- The influence of equivalence ratio is negligible
- Ignition delay times decrease with increasing pressure

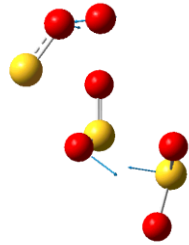
*Base mechanism*



*Modified mechanism*



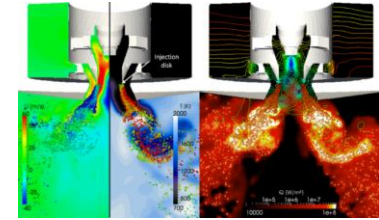
# Burner design and sulphur atomization



Combustion  
kinetics  
development



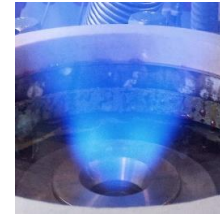
CFD  
Simulations



Sulphur  
Atomization  
(lab scale)

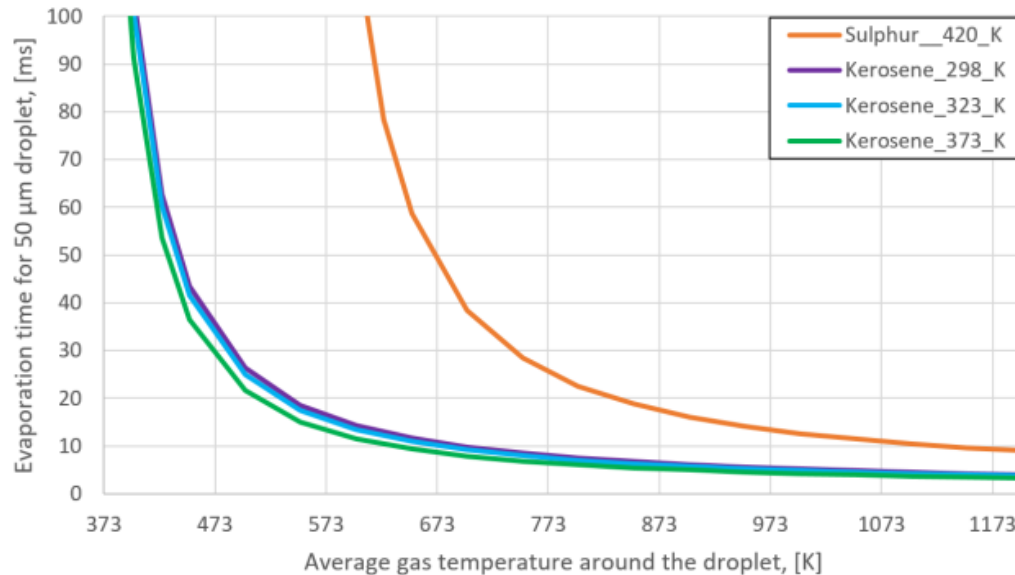


Sulphur  
Combustion  
(lab scale)



# Atomization is key factor in burner design

Sulphur droplets require more time for evaporation compared to typical liquid fuels



# Atomizer selection

Performance of 4 types of atomizers for sulphur atomization were analyzed by correlations based on key parameters for atomization ( $We$ ,  $Oh$ ,  $Re$ ) → **Sauter Mean Diameter (SMD)**

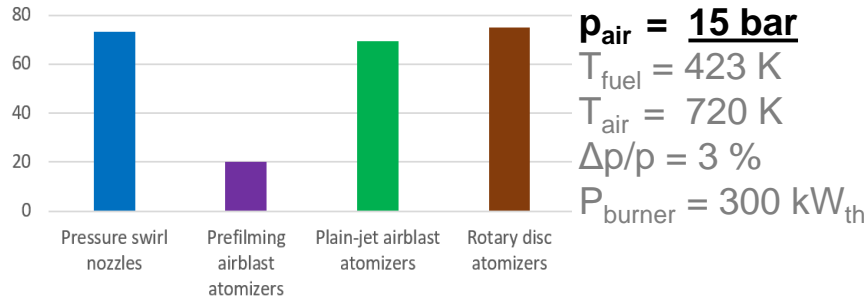
SMD can be defined as the diameter of a drop having the same **volume/surface** area ratio as the entire spray.

- **Pressure swirl atomizer**
- **Prefilming airblast atomizer**
- **Plain-jet airblast orifice diameter**
- **Rotary atomizer**

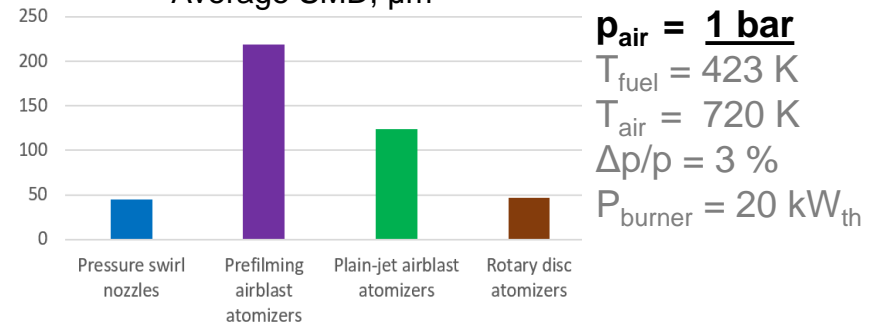
Best performance in laboratory scale (right)

Best performance for real scale gas turbine conditions (left)

Average SMD,  $\mu\text{m}$



Average SMD,  $\mu\text{m}$

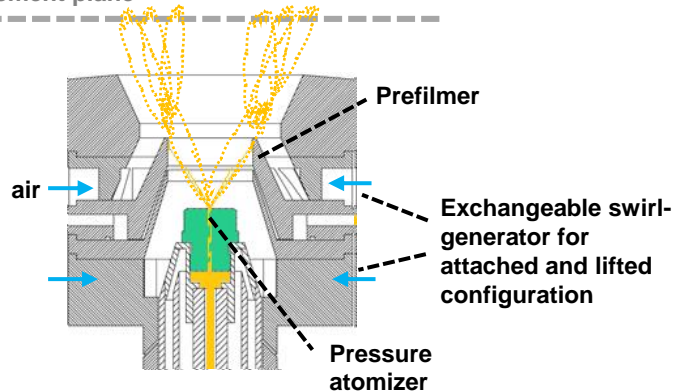


# Burner design

## Prototypes for experiments

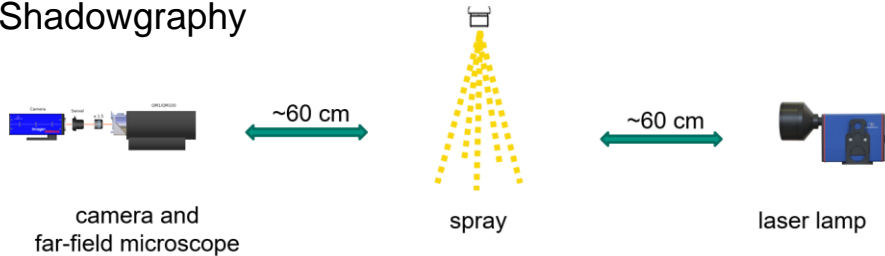
- Can be operated with pressure atomizer or prefilming airblast atomizer
- Modular burner construction allows to investigate lifted and attached flame configuration by changing swirl generators

Spray measurement plane

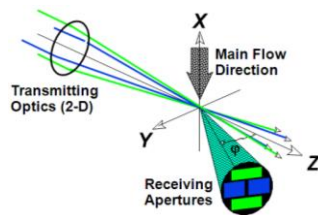


# Experimental setup - spray test rig

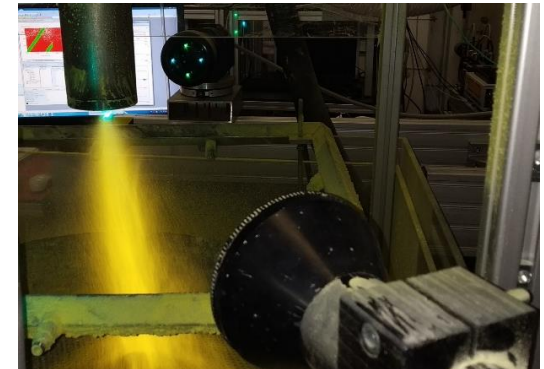
- Liquid sulphur supply at up to 200 bar, 413 K
- Inlet air temperature set to 413 K
- Measurement techniques:
  - Shadowgraphy



- Phase Doppler Anemometry (PDA)

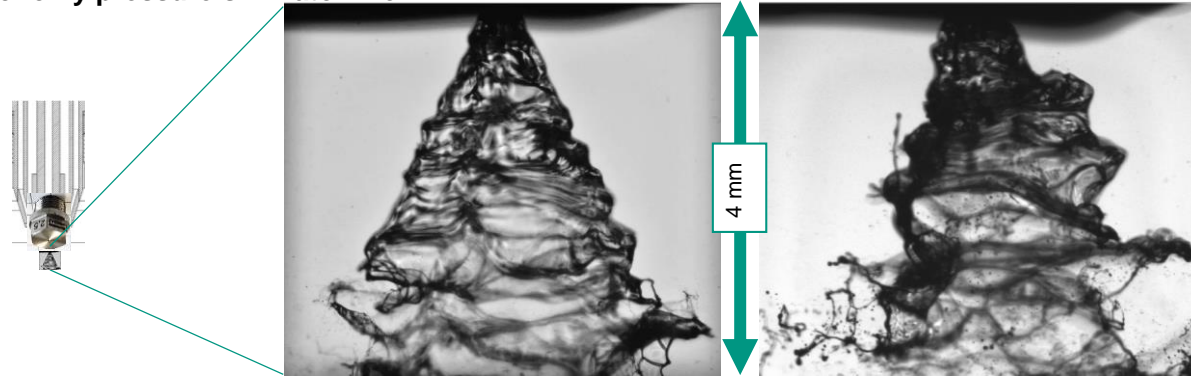


PDA principle, Dantec



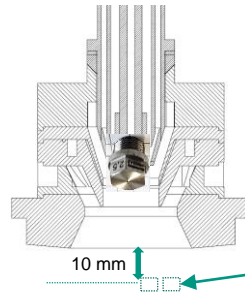
# Sulphur spray results

- Shadowgraphy with far field microscope: Qualitative analysis of primary atomization  
Operation of only pressure swirl atomizer

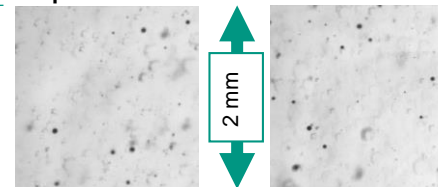


Complete burner nozzle (with swirled air)

Cone angle  $\sim 70^\circ$  (Hollow cone)



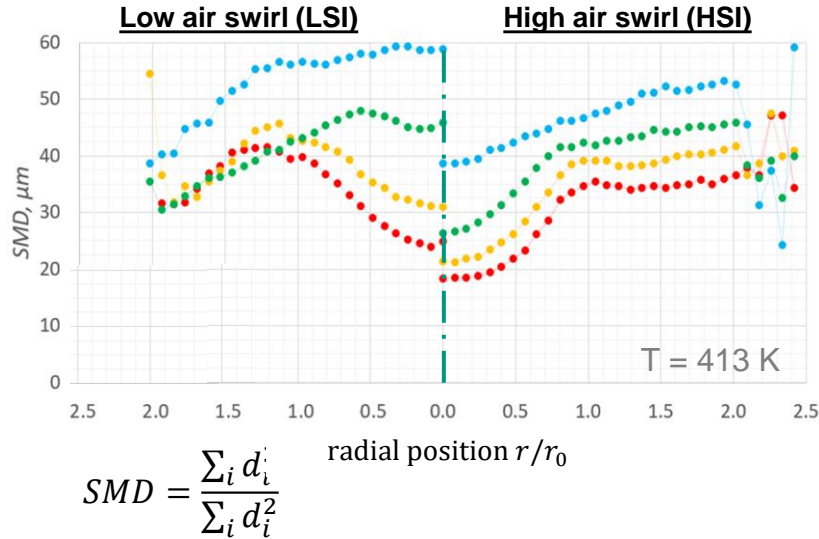
Primary atomization is completed at selected measurement plane for PDA measurements





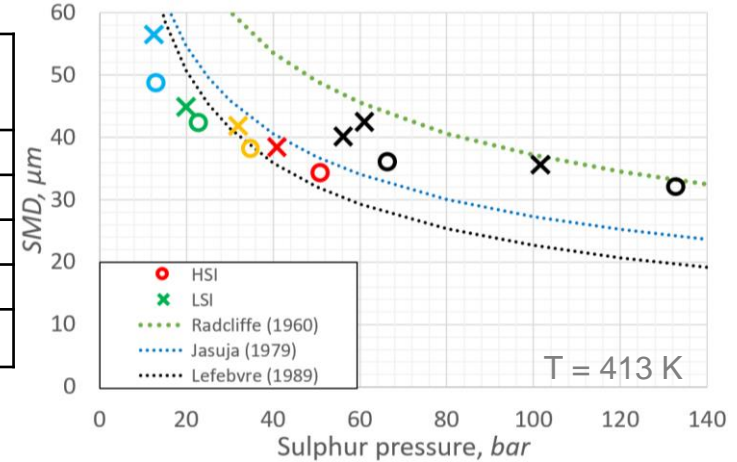
# Sulphur spray – Sauter Mean Diameter (SMD)

SMD for different type of air flow fields



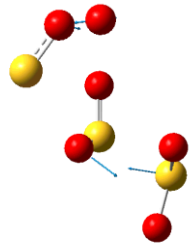
Eq. thermal power [kW]	Sulphur atomization pressure [bar]
22.5	46
20.0	33
17.0	22
13.5	13
>22.5	>46 (richer mixtures)

Global SMD vs atomization pressure



➡ Measured particle size distribution is used as input to CFD simulations

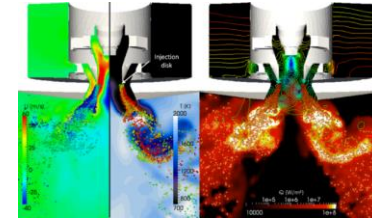
# CFD simulations



Combustion kinetics development



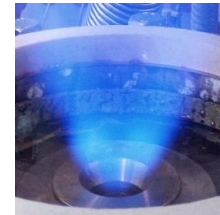
CFD Simulations



Sulphur Atomization (lab scale)



Sulphur Combustion (lab scale)

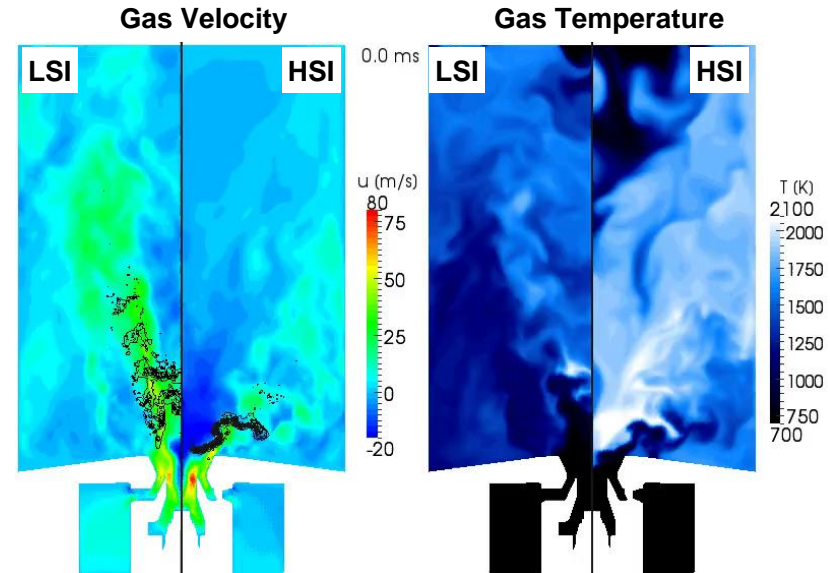
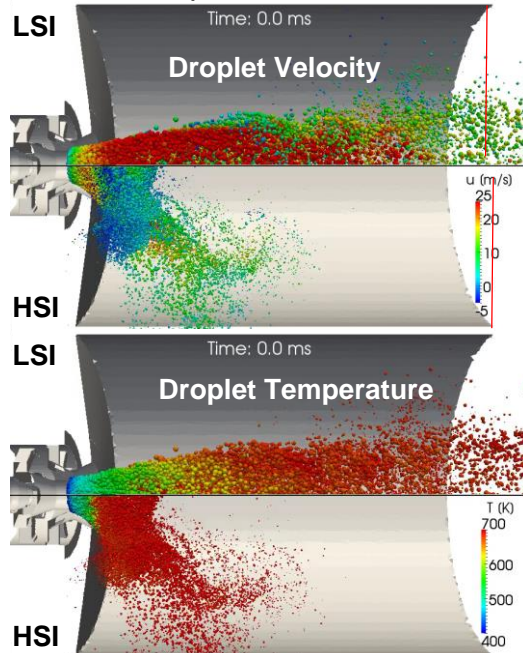


# Low and High Swirl Intensity Burner (LSI vs HSI)

- LES simulation (8 million cells) with OpenFOAM
  - Lagrangian-Eulerian method for multiphase
  - PaSR [- partially stirred Reactor] combustion model based on sulphur kinetics

	Inlet $D_{32}$ [ $\mu\text{m}$ ]	$L_p$ [mm] - (95%)	Evap	$\dot{Q}$ [kW]
LSI	50	148	99.4%	19.7
HSI	50	90	100%	19.1

(Combustor air inlet temperature 720 K, combustor pressure 1 bar)



# Evaluation of two Nozzle Designs

## HSI

- Compact reaction zone
- High local flame temperature
- High  $\text{SO}_2$ , low  $\text{SO}_3$
- High local  $\Phi$

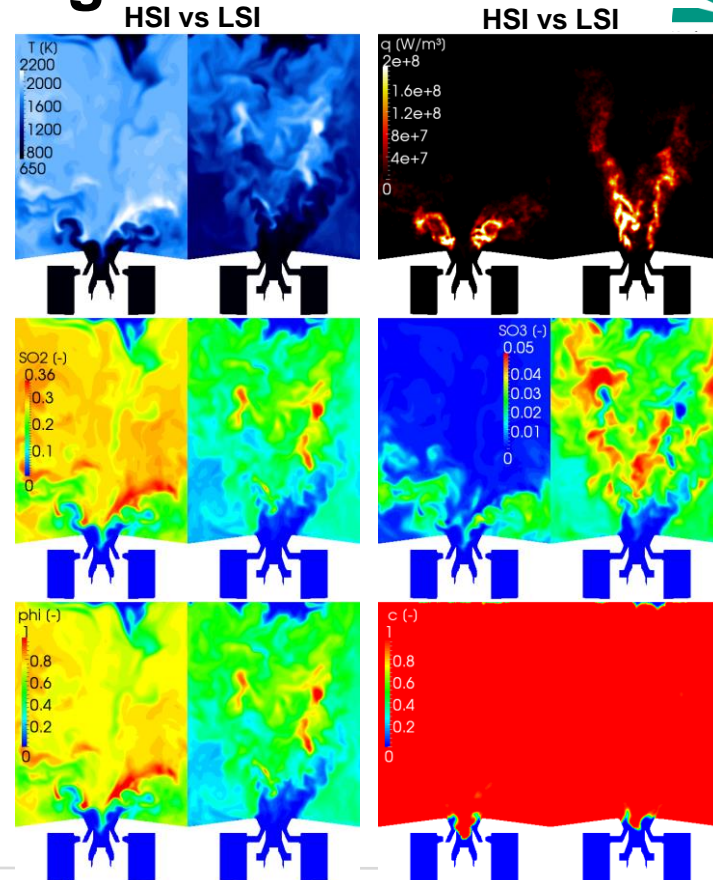
## LSI

- Long narrow reaction zone
- Low local flame temperature
- Low  $\text{SO}_2$ , high  $\text{SO}_3$
- Low local  $\Phi$



**LSI Burner Preferred**

Next step: high pressure operations



# Pilot scale design - elevated pressure

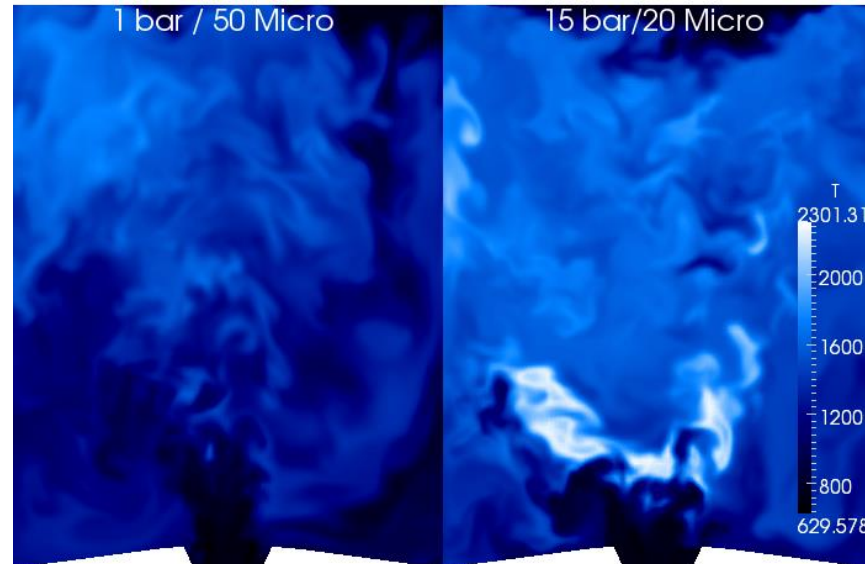
## Combustion laboratory scale:

- Pressure: 1 bar
- Thermal load ca. 20 kW
- Power density over combustion chamber volume: 5 MW/m<sup>3</sup>  
(target: > 1.5 MW/m<sup>3</sup>)

## Combustion at elevated pressure:

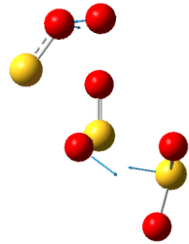
- Pressure: 15 bar
- Thermal load ca. 300 kW
- Power density over combustion chamber volume: 72 MW/m<sup>3</sup>

## Comparison at 1 and 15 bar combustor pressure



Higher overall combustion temperature at elevated pressure

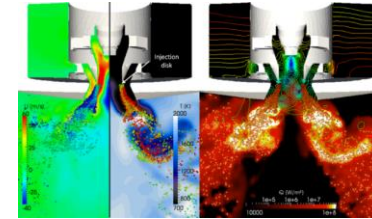
# Sulphur combustion (lab scale)



Combustion kinetics development



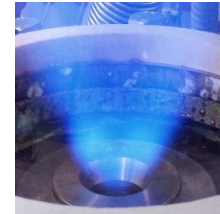
CFD Simulations



Sulphur Atomization (lab scale)



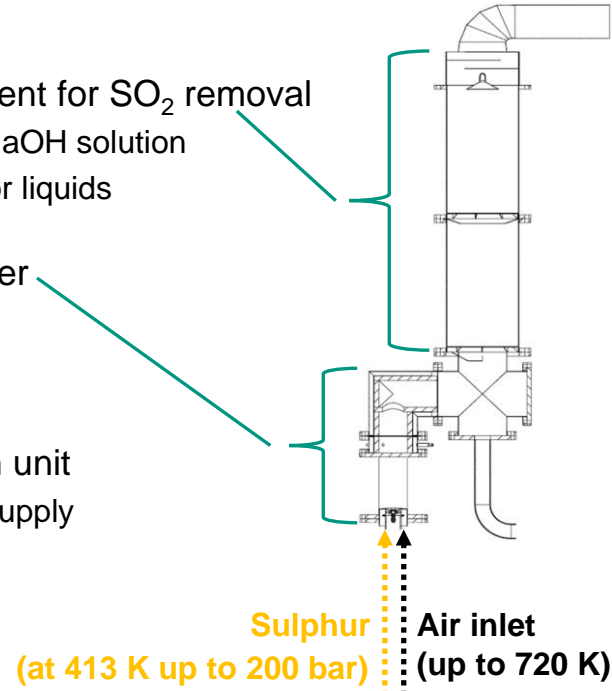
Sulphur Combustion (lab scale)



# Experimental setup - Combustion test rig

## ■ Components

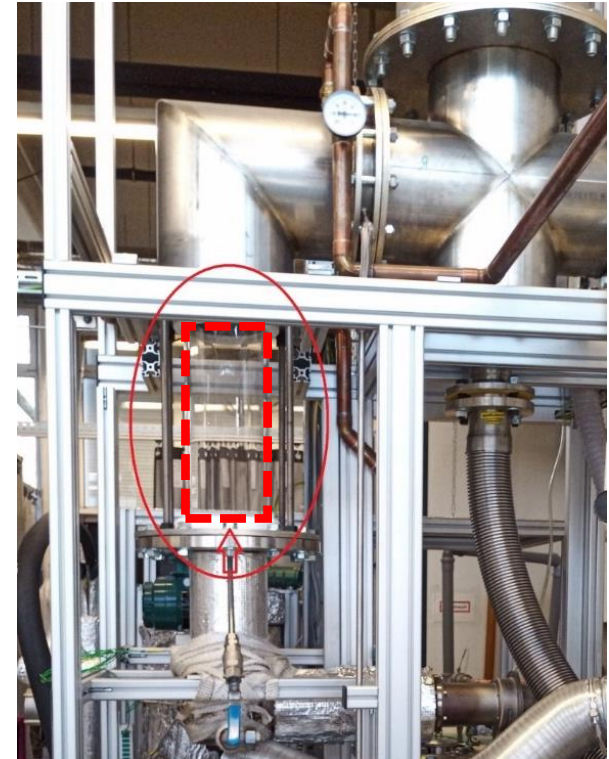
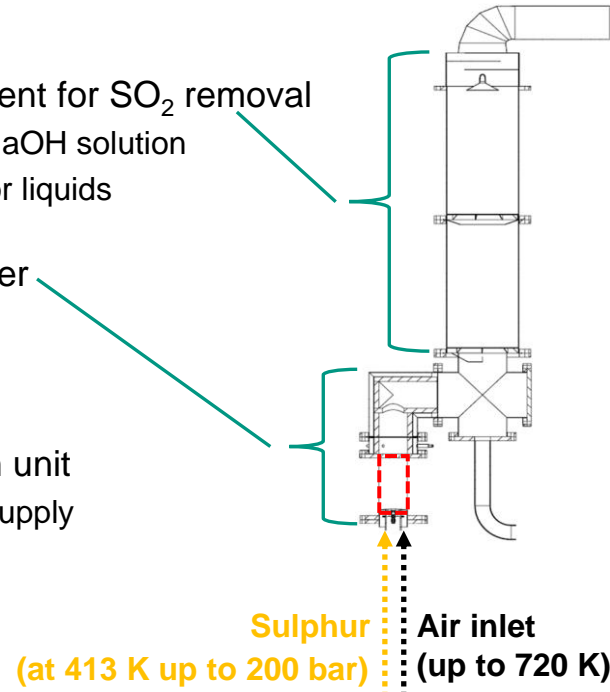
- Exhaust gas treatment for SO<sub>2</sub> removal
  - Scrubber with NaOH solution
  - Closed circuit for liquids
- Combustion chamber
  - Optical access
  - Ceramic inlays
- Sulphur preparation unit
  - Liquid sulphur supply



# Experimental setup - Combustion test rig

## ■ Components

- Exhaust gas treatment for SO<sub>2</sub> removal
  - Scrubber with NaOH solution
  - Closed circuit for liquids
- Combustion chamber
  - Optical access
  - Ceramic inlays
- Sulphur preparation unit
  - Liquid sulphur supply



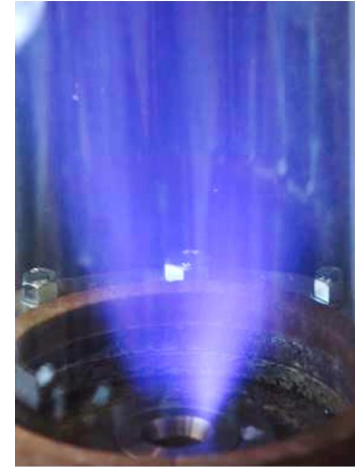


# Combustion experiments

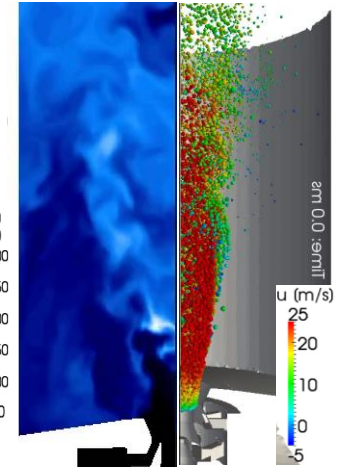
## Results for LSI configuration

- Ignition by auto ignition (air inlet temperature 720 K)
- Power density > 5 MW/m<sup>3</sup> (ambient pressure)  
target: > 1.5 MW/m<sup>3</sup>
- Very large stability range  
(also for low air inlet temperatures 367 K)

## LSI configuration



16 kW (air inlet temperature=720 K)



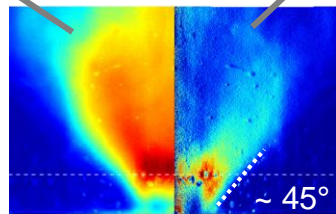
CFD simulation



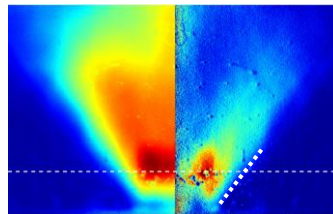
# Flame structure in LSI configuration

Line-of-sight

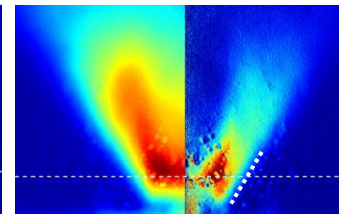
Abel-transformed



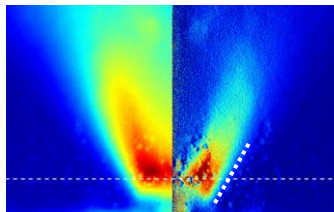
$P = 21.4 \text{ kW}$   
 $\lambda = 1.56$   
 $P_S = 110 \text{ bar}$



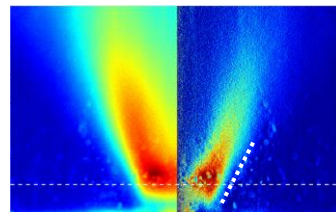
$P = 20.7 \text{ kW}$   
 $\lambda = 1.62$   
 $P_S = 101 \text{ bar}$



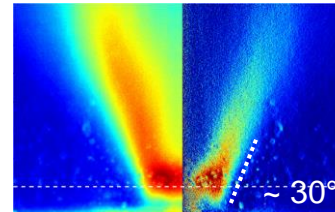
$P = 18.1 \text{ kW}$   
 $\lambda = 1.85$   
 $P_S = 87 \text{ bar}$



$P = 15.5 \text{ kW}$   
 $\lambda = 2.16$   
 $P_S = 62 \text{ bar}$



$P = 12.9 \text{ kW}$   
 $\lambda = 2.59$   
 $P_S = 43 \text{ bar}$

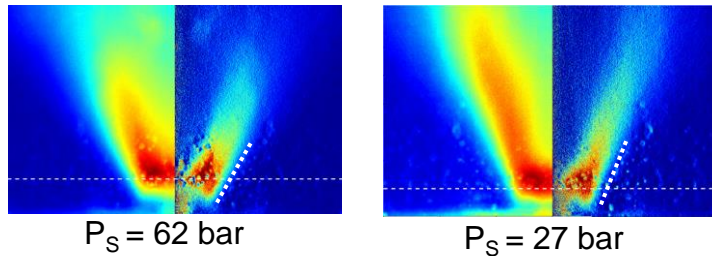


$P = 10.3 \text{ kW}$   
 $\lambda = 3.24$   
 $P_S = 27 \text{ bar}$

$(\Delta p/p_{\text{air}} = 3 \%, T_{\text{inlet}} = 450 \text{ }^\circ\text{C})$

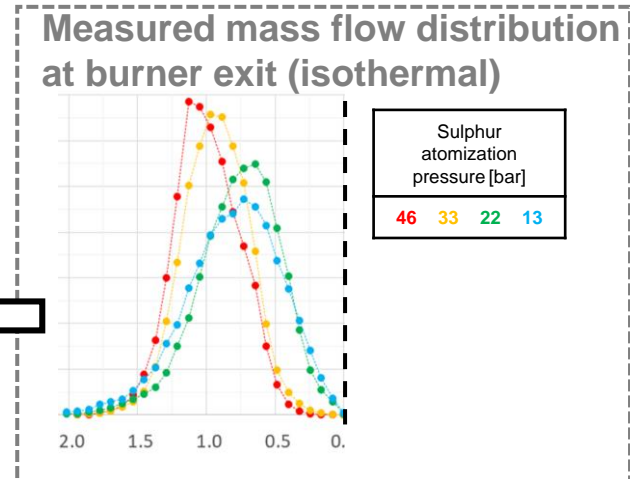
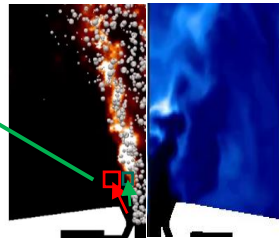
# Flame structure in LSI configuration

- Flame shape is dependent on Sulphur atomization pressure



- Initial velocity vectors of spray in CFD simulations have to be adapted to allow precise simulations for varied operating conditions

spray is more narrow compared to experiment



# Conclusions

- **Refined CFD simulations for burner development based on**
  - Developed sulphur kinetics
  - Atomization of sulphur at relevant conditions
  
- **High power density combustion seems feasible**
  - Lab scale experiments and simulations show  
**Power density > 5 MW/m<sup>3</sup> at ambient pressure (target > 1.5 MW/m<sup>3</sup>)**
  - Burner works well for elevated pressure conditions based on simulations  
**Power density > 72 MW/m<sup>3</sup> at 15 bar - gas turbine condition**
  
- **More research required:**
  - Measurement of laminar flame speed & auto ignition times
  - Detailed validation of kinetics
  - Burner upscaling and optimization
  - Study on combustor and turbine blade materials for sulphur combustion



# Acknowledgements

## Thank you

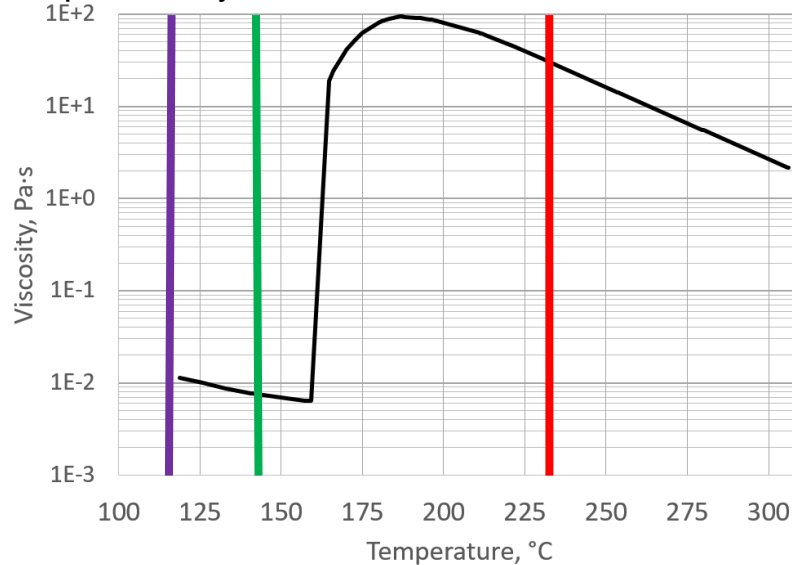


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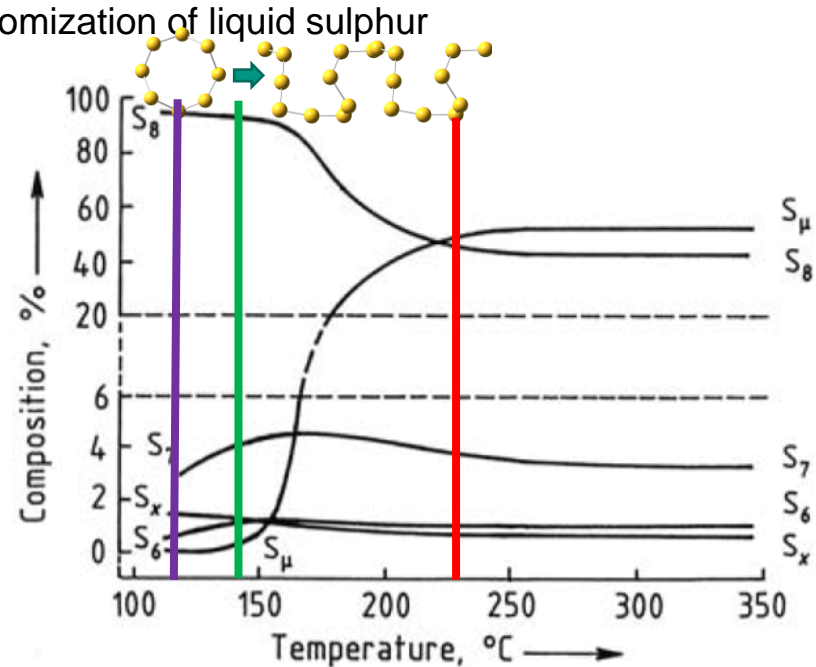


# Sulphur - properties

- Viscosity of liquid sulphur strongly dependent on temperature due to polymerization
- Step viscosity increase shall be avoided before atomization of liquid sulphur



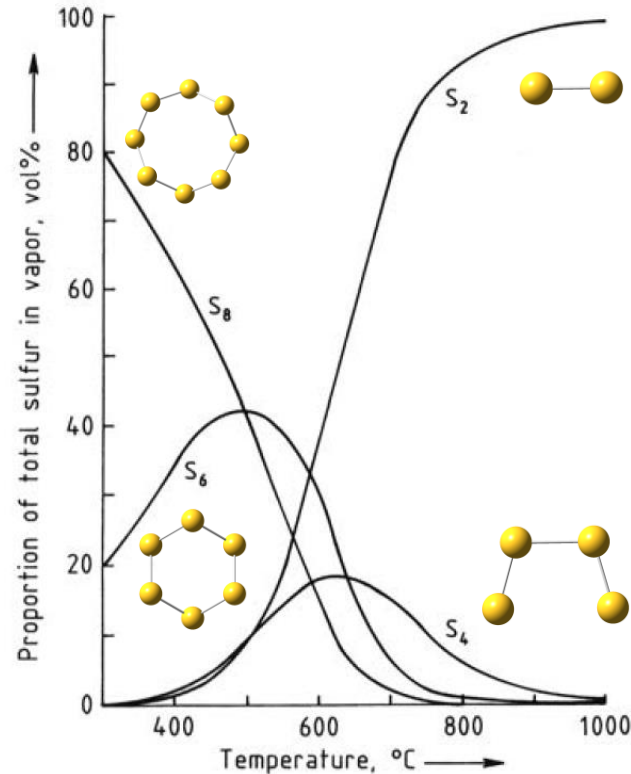
- 115 °C – Melting temperature
- 140 °C – Temperature of liquid sulphur supply
- 230 °C – Autoignition temperature
- 450 °C – Air inlet temperature of gas turbine combustor



Source: W. NEHB, K. VYDRA (ULLMANN'S encyclopedia of industrial chemistry)

# Sulphur - properties

## ■ Composition of sulphur vapor



Source: W. NEHB, K. VYDRA  
(ULLMANN'S encyclopedia of industrial  
chemistry)

# Sulphur - properties

Parameter	Unit	Liquid sulphur	Kerosene, Jet-A1
		@ 423 K	@ 298 K
Density	kg/m <sup>3</sup>	1780	810
Surface tension	mN/m	61	26
Viscosity	mPa·s	7.0	1.5
Specific heat	kJ/(kg·K)	1.1	2.0
Heat of vaporization	kJ/kg	290	363
Heat of combustion (LHV for kerosene)	kJ/kg	-9 300 (for solid)	-43 000
Vapor pressure	Pa	26	185



# Particle size distribution

- Measured particle size distribution is used as input to CFD simulations as Rosin-Rammler distribution

